

Satellite Overview

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Slides 13



Wireless Networks



– Wireless Wide Area Networks (WWANs)

- Cellular Networks :
 - GSM, cdmaone (IS-95), UMTS, cdma2000 EVDO
- Satellite Systems:
 - Iridium, Inmarsat, GPS, etc.



– Wireless Metro Area Networks (WMANs)

- IEEE 802.16 WiMAX

– Wireless Local Area Networks (WLANs)

- IEEE 802.11, a, b, g, etc. (infrastructure, ad hoc)

– Wireless Personal Area Networks (WPANs)

- IEEE 802.15 (Bluetooth), IrDa, Zigbee, sensor, etc.

Satellite Applications



- Over 3500 Satellites in use today - overall industry revenues several billion in 2007
- Telecommunications
 - global telephone connections
 - backbone for global networks
 - connections for communication in remote places or underdeveloped areas
 - global mobile communications
- Other Applications
 - Weather
 - Radio and TV broadcast satellites
 - Earth observation (climate change, agricultural, etc.)
 - Military: surveillance, imaging, intelligence, early warning
 - Navigation and localization: aeronautic, nautical, etc., (e.g., GPS)



largely replaced by fiber optics

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Satellite History



A Selective Satellite Chronology

- 1945 Arthur C. Clarke – head of British Interplanetary Society publishes article: "Extra-Terrestrial Relays" defines basic satellite concept - fame as Sci-Fi author (2001 A Space Odyssey)
- 1955 John R. Pierce (AT&T researcher) publishes Article: "Orbital Radio Relays" defines technical aspects for global telephony
- 1957 Sputnik: USSR launches the first earth satellite.
- 1960 First reflecting communication satellite ECHO – aluminum coated balloon
- 1962 AT&T TELSTAR (first active satellite) and RELAY launched (MEO) - for voice/television/data
- 1963 SYNCOM launched – first geosynchronous orbit satellite



<http://www.hq.nasa.gov/office/pao/History/satcomhistory.html>

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Satellite History



- 1964 INTELSAT formed
International Telecommunications Satellite Organization (part of UN) - aim provide global telecommunications connectivity
- 1965 COMSAT's EARLY BIRD: 1st commercial communications satellite: 240 duplex telephone channels or 1 TV channel, 1.5 years lifetime
- 1969 INTELSAT-III series provides global coverage
- 1972 ANIK: 1st Domestic Communications Satellite (Canada)
- 1974 WESTAR: 1st U.S. Domestic Communications Satellite
- 1976 MARISAT: 3 communications satellites for maritime communications
- 1979 INMARSAT formed.
- 1988 first satellite system for mobile phones and data communication INMARSAT-C
- 1993 first digital satellite telephone system
- 1998 global satellite systems for small mobile phones



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Satellite System Components

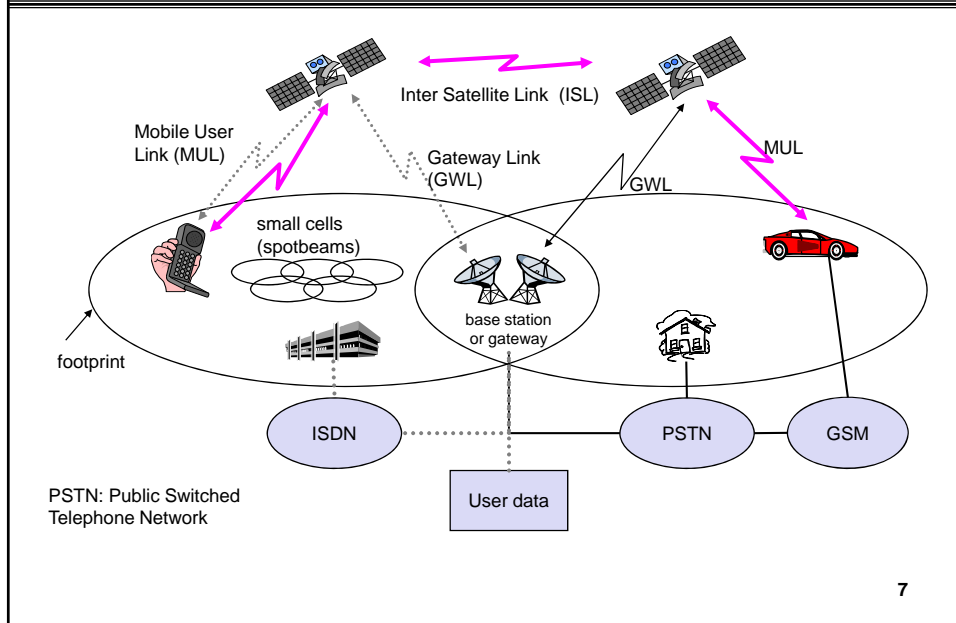


- Earth Stations (Ground Stations) – antenna systems on or near earth
- Uplink – transmission from an earth station to a satellite
- Downlink – transmission from a satellite to an earth station
- Typically separated frequencies for uplink and downlink (FDD)
- Transponder – electronics in the satellite that convert uplink signals to downlink signals
 - transparent transponder: only shift frequencies (Bent Pipe)
 - regenerative transponder: additionally signal regeneration and formatting) Processing Satellite



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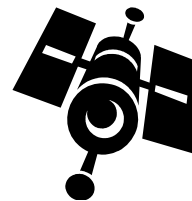
Typical Satellite System



Satellite Classification



- **Frequency:**
 - Spectrum utilized – controlled by ITU
- **Orbit:**
 - Height: (GEO, HEO, MEO, LEO)
 - Pattern: elliptical vs. circular, inclination, etc.
- **Multiple Access Method:**
 - TDMA, FDMA, CDMA
- **Satellite Capabilities**
 - Bent Pipe vs. On board Processing
- **Coverage and Usage Type**
 - Coverage: global, regional, national
 - Usage
 - Fixed service satellite (FSS)
 - Broadcast service satellite (BSS)
 - Mobile service satellite (MSS)



U.S. Satellite Frequency Bands



Licensed Spectrum - only

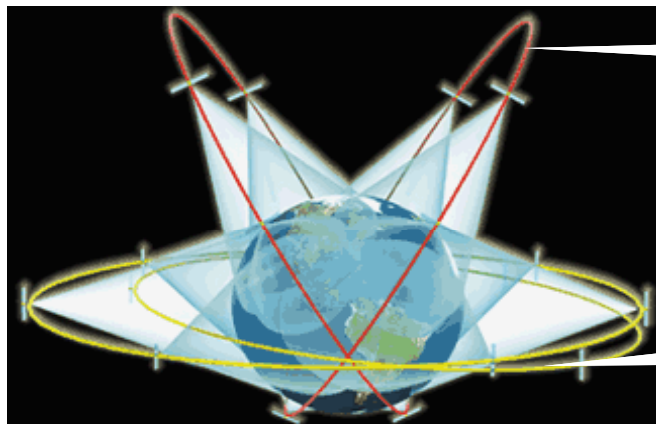
Band	Frequency Range	Total Bandwidth	General Application
L	1 to 2 GHz	1 GHz	Mobile satellite service (MSS)
S	2 to 4 GHz	2 GHz	MSS, NASA, deep space research
C	4 to 8 GHz	4 GHz	Fixed satellite service (FSS)
X	8 to 12.5 GHz	4.5 GHz	FSS military, terrestrial earth exploration, and meteorological satellites
Ku	12.5 to 18 GHz	5.5 GHz	FSS, broadcast satellite service (BSS)
K	18 to 26.5 GHz	8.5 GHz	BSS, FSS
Ka	26.5 to 40 GHz	13.5 GHz	FSS

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Classification of Satellite Orbits



- Circular or elliptical orbit
 - Circular with center at earth's center
 - Elliptical with one foci at earth's center



Elliptical

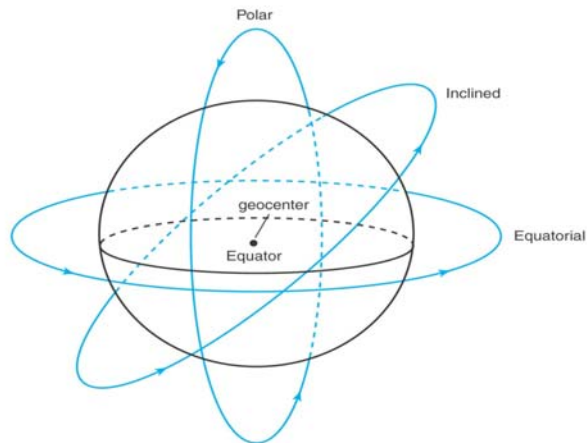
Circular

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Orbital Pattern



- Orbit around earth in different planes
 - Equatorial orbit above earth's equator
 - Polar orbit passes over both poles
 - Other orbits referred to as inclined

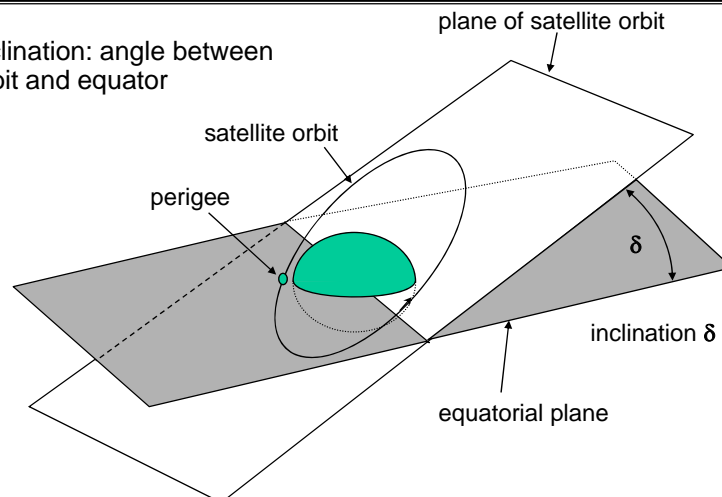


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Inclination



inclination: angle between orbit and equator

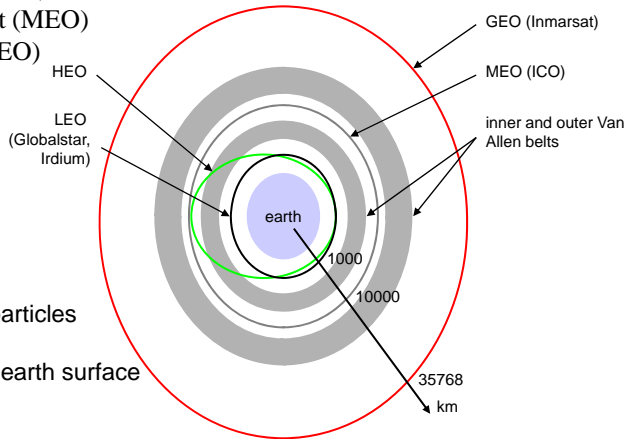


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Classification of Satellite Orbits

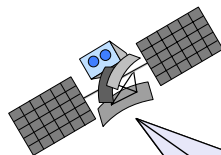


- Altitude of satellites
 - Geostationary orbit (GEO)
 - High earth orbit (HEO)
 - Medium earth orbit (MEO)
 - Low earth orbit (LEO)



Van-Allen-Belts: ionized particles
 2000 - 6000 km and
 15000 - 30000 km above earth surface

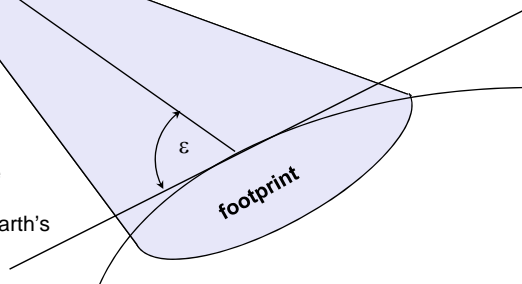
Elevation



Elevation:
 angle ϵ between center of satellite beam
 and surface

minimal elevation:
 elevation needed at least
 to communicate with the satellite

Coverage Angle: portion of the earth's
 Surface visible to satellite



Satellite Footprint



Footprint: geographic area where communication possible

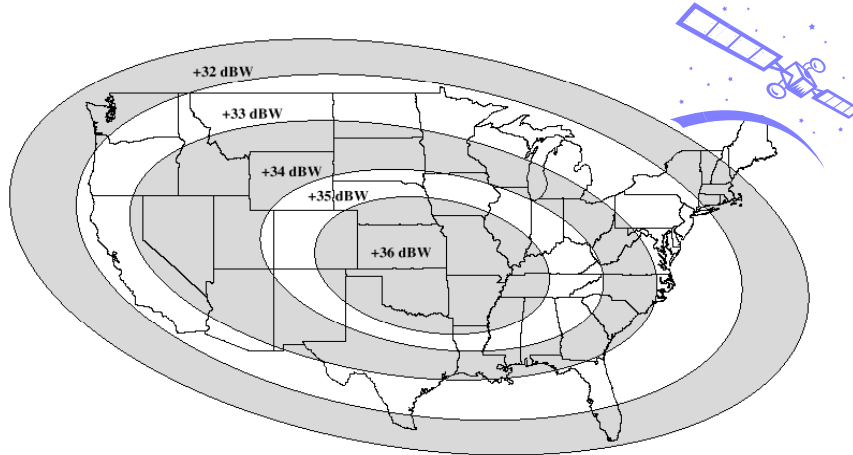


Figure 9.6 Typical Satellite Footprint

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Satellite Footprint



- Example INTELSAT
- Horizons -2 74°W

Ku-Band N. America Beam Peak up to 53.6 dBW



Ku-Band Europe/N. America Beam Peak up to 55.0 dBW



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Link budget of satellites

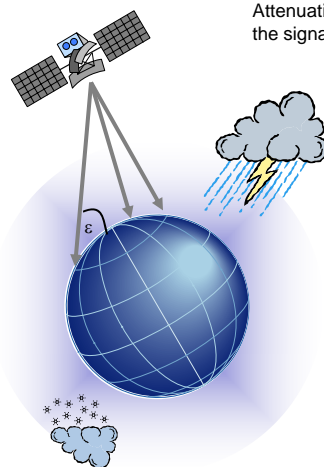


- Parameters like attenuation or received power determined by four parameters:
 - Transmit power P_t
 - Gain of sending antenna G_t
 - Distance between sender and receiver d
 - Gain of receiving antenna G_r
- Path Loss is modeled well by free space model
 - Path Loss = $PL = 21.98 - 20\log_{10}(\lambda) + 20\log_{10}(d)$
- $P_r = P_t + G_t + G_r - PL$
- Problems
 - interruptions due to shadowing of signal (no Line of Sight NLOS)
 - Typically need LOS in satellite systems
 - High frequencies are used and are subject to atmospheric effects
 - Possible solutions
 - Link Margin to eliminate variations in signal strength
 - satellite diversity (usage of several visible satellites at the same time) helps to use less transmit power



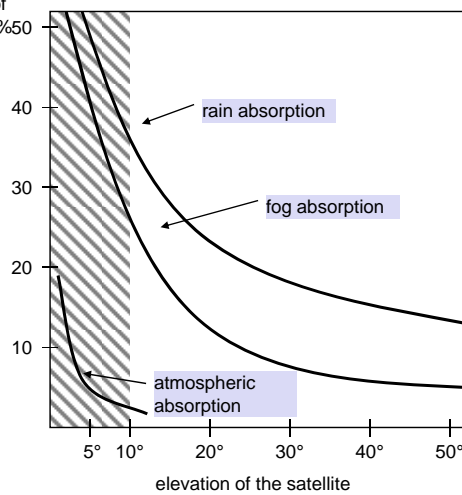
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Atmospheric attenuation



Attenuation of the signal in %50

Example: satellite systems at 4-6 GHz



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Orbit Basics



- Satellites in circular orbits
 - attractive force $F_g = m g (R/r)^2$
 - centrifugal force $F_c = m r \omega^2$
 - m : mass of the satellite
 - R : radius of the earth ($R = 6370$ km)
 - r : distance to the center of the earth
 - g : acceleration of gravity ($g = 9.81$ m/s²)
 - ω : angular velocity ($\omega = 2 \pi f$, f : rotation frequency)

- Stable orbit

- $F_g = F_c$

$$r = \sqrt[3]{\frac{gR^2}{(2\pi f)^2}}$$

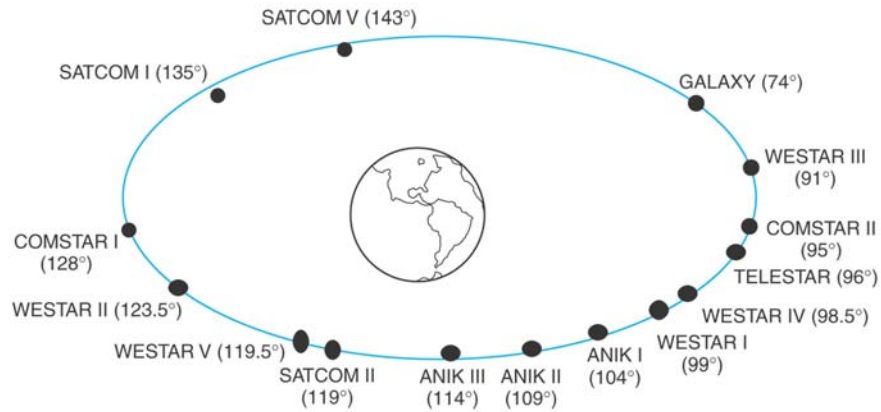
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Geostationary satellites



- Orbit 35.786 km distance to earth surface, orbit in equatorial plane (inclination 0°)
- → complete rotation exactly one day, satellite is synchronous to earth rotation
- Advantages of the GEO orbit
 - No problem with frequency changes
 - Tracking of the satellite is simplified
 - High coverage area with large footprint (up to 34% of earth surface!),
- Disadvantages of the GEO orbit
 - Weak signal after traveling over 35,000 km → high transmit power
 - Bad elevations in areas with latitude above 60° due to fixed position above the equator
 - Polar regions are poorly served
 - high latency due to long distance (~ 275 ms)
- Geosynchronous orbit – now called Clarke orbit

Satellites in geosynchronous earth orbits

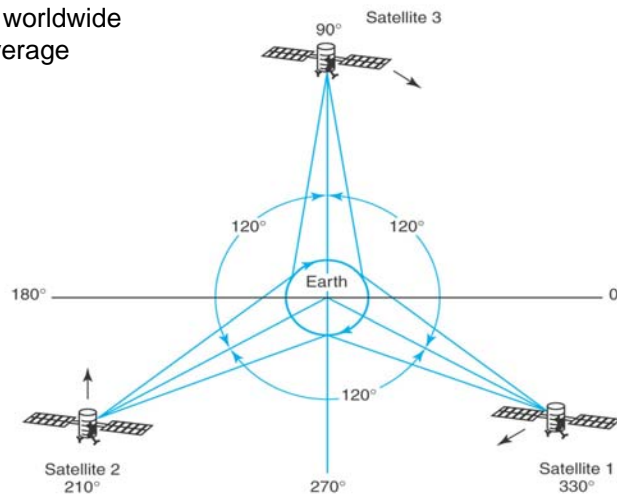


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Three geosynchronous satellites in Clarke orbits



3 satellites can provide worldwide coverage



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GEO Systems



- Example
- INMARSAT – GEO system originally for maritime communication
 - handheld phone not available due to high power required
 - focus now on data service as well as telephony

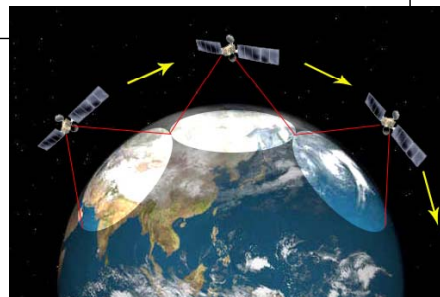
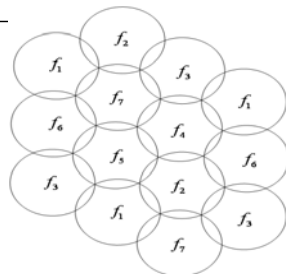


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LEO systems



- Orbit 500 - 1500 km above earth surface
 - visibility of a satellite 10 - 40 minutes
 - latency comparable with terrestrial long distance connections, ca. 5 - 10 ms
 - many satellites necessary for global coverage
 - handover necessary from one satellite to another
 - more complex systems due to moving satellites
 - Note satellites use spot beams with smaller footprints - allows frequency reuse



LEO Systems



- Examples:
 - Iridium (start 1998, 66 satellites in 11 orbits – 6 in each orbit)
 - Smart system with inter satellite links (ISL) links
 - Globalstar (start 1999, 48 satellites in 8 orbits – 6 in each orbit)
 - Bent pipe system



GlobalStar



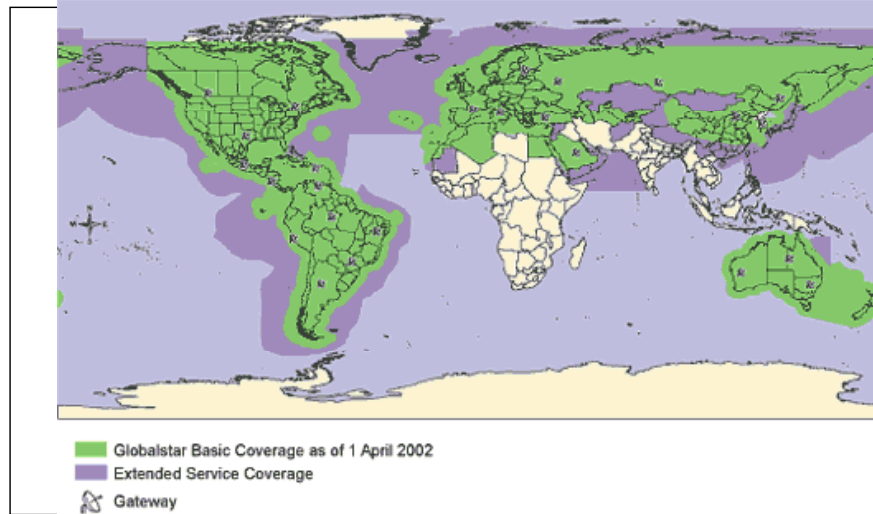
- Intended as adjunct to cellular
 - Dual Mode, dual band phones
 - IS95 – Globalstar,
 - Phone first tries to find cellular service
 - If not available – home on to satellite
 - LEO system 48 satellites
 - 8 orbital planes – 6 satellites each
 - Cheap Bent pipe system



QUALCOMM GSP-1600



Global Star



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Routing Connections



- In LEO networks need to route connections through the network
- Some LEOs use inter satellite links (ISL) (e.g., Iridium)
 - reduced number of gateways needed
 - forward connections or data packets within the satellite network as long as possible
 - only one uplink and one downlink per direction needed for the connection of two mobile phones or data connections
- Problems:
 - more complex focusing of antennas between satellites
 - high system complexity due to moving routers
 - higher fuel consumption
 - thus shorter lifetime
- Other systems use multiple gateways on the earth and additionally terrestrial networks (e.g, GlobalStar – uses cellular networks when in range)

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Localization of mobile stations



- Mechanisms similar to GSM
- Gateways maintain registers with user data
 - HLR (Home Location Register): static user data
 - VLR (Visitor Location Register): (last known) location of the mobile station
 - SUMR (Satellite User Mapping Register):
 - satellite assigned to a mobile station
 - positions of all satellites
- Registration of mobile stations
 - Localization of the mobile station via the satellite's position
 - requesting user data from HLR
 - updating VLR and SUMR
- Calling a mobile station
 - localization using HLR/VLR similar to GSM
 - connection setup using the appropriate satellite

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MEO systems



- Orbit ca. 5000 - 12000 km above earth surface
- comparison with LEO systems:
- slower moving satellites
- Diameter of coverage is 10,000 to 15,000 km
 - less satellites needed
 - simpler system design
 - for many connections no hand-over needed
- higher latency, ca. 70 - 80 ms
- higher sending power needed
- special antennas for small footprints needed
- Orbit period of 6 hours or more
- Maximum satellite visible time is a few hours
- Example:
- Ellipso/ICO



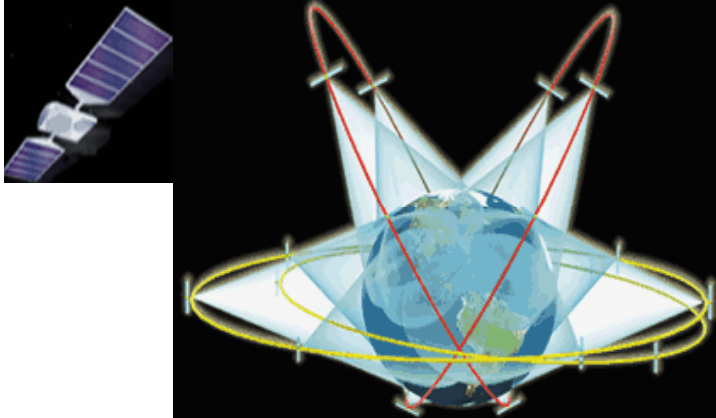
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Ellipso System



- Set of 17 satellites to provide worldwide coverage with dual mode cellular/satellite phones

Satellite to mobile 2165 - 2200 MHz S band
Mobile to satellite 1990 - 2025 MHz S band
Satellite to gateway 6775 - 7075 MHz C band
Gateway to satellite 15400-15700 MHz Ku band

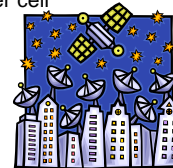


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Handover in satellite systems



- Several additional situations for handover in satellite systems compared to cellular terrestrial mobile phone networks - handoffs caused by the movement of the satellites
 - Intra satellite handover
 - handover from one spot beam to another
 - mobile station still in the footprint of the satellite, but in another cell
 - Inter satellite handover
 - handover from one satellite to another satellite
 - mobile station leaves the footprint of one satellite
 - Gateway handover
 - Handover from one gateway to another
 - mobile station still in the footprint of a satellite, but gateway leaves the footprint
 - Inter system handover
 - Handover from the satellite network to a terrestrial cellular network
 - mobile station can reach a terrestrial network again which might be cheaper, has a lower latency etc.



Overview of LEO/MEO systems



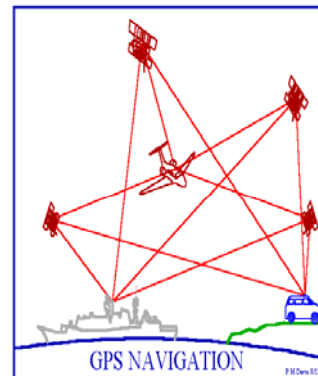
	Iridium	Globalstar	Ellipso/ICO	Teledesic
# satellites	66 + 6	48 + 4	17 + 2	288
altitude (km)	780	1414	10390	ca. 700
coverage	global	±70° latitude	global	global
min. elevation	8°	20°	20°	40°
frequencies [GHz] (circa)	1.6 MS 29.2 ↑ 19.5 ↓ 23.3 ISL	1.6 MS ↑ 2.5 MS ↓ 5.1 ↑ 6.9 ↓	2 MS ↑ 2.2 MS ↓ 5.2 ↑ 7 ↓	19 ↓ 28.8 ↑ 62 ISL
access method	FDMA/TDMA	CDMA	FDMA/TDMA	FDMA/TDMA
ISL	yes	no	no	yes
bit rate	2.4 kbit/s	9.6 kbit/s	4.8 kbit/s	64 Mbit/s ↓ 2/64 Mbit/s ↑
# channels	4000	2700	4500	2500
Lifetime [years]	5-8	7.5	12	10
cost estimation	4.4 B\$	2.9 B\$	4.5 B\$	9 B\$
Access	TDD/TDMA	FDD/CDMA	CDMA	FDMA/TDMA

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Global Position System



- In 1973, the US Air Force proposed a new system for navigation using MEO satellites
- The system is known as: *Navigation System with Timing and Ranging: Global Positioning System* or *NAVSTAR GPS*
- Full Operational Capability declared by the Secretary of Defense at 00:01 hours on July 17, 1995
- Goals
 - What time is it?
 - What is my position (including attitude)?
 - What is my velocity?
 - What is the distance between two points?
 - What is my estimated time of arrival?

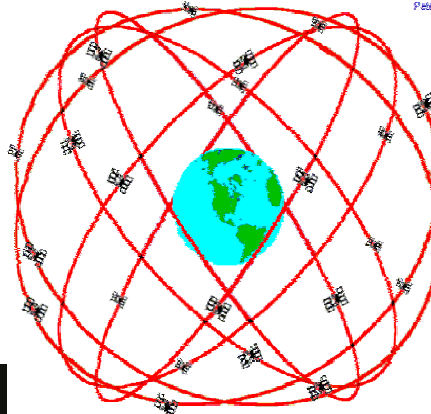
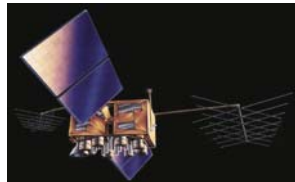


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Space Segment



- System consists of 24 satellites: 21 in use and 3 spares
- Altitude: 20,200 Km with periods of 12 hr (MEOs).
- Hydrogen maser atomic clocks - lose one second every 2,739,000 million years



GPS Nominal Constellation
24 Satellites in 6 Orbital Planes
4 Satellites in each Plane
20,200 km Altitudes, 55 Degree Inclination

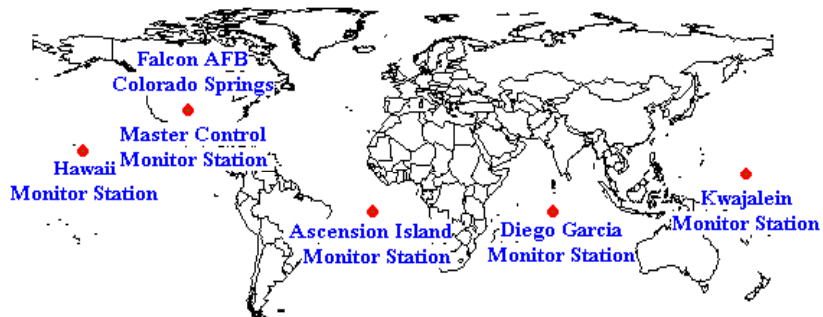
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Control Segment



- Master Control Station is located at the Consolidated Space Operations Center (CSOC) at Falcon Air Force Station near, Colorado Springs, Colorado
- Tracks the satellites for orbit and clock determination

Peter H. Dana 5/27/95

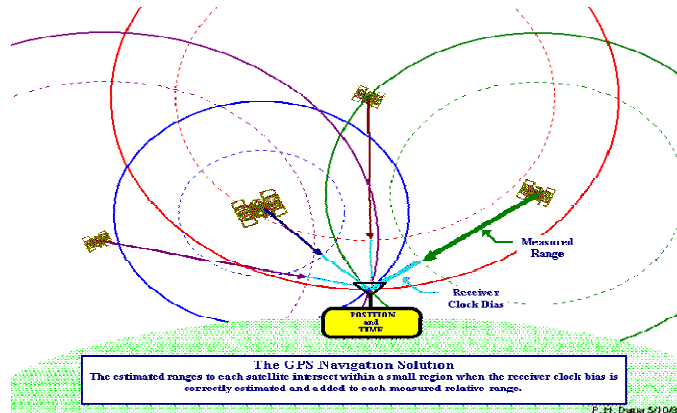


Global Positioning System (GPS) Master Control and Monitor Station Network

GSP System: Overview



- Satellites simultaneously broadcast beacon messages (called navigation messages)
- A GPS receiver measures time of arrival from the satellites, and then uses "trilateration" to determine its position
- 4th satellite improves accuracy as satellite clocks not in synch



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GPS Satellite Transmissions



- All 24 GPS satellites transmit on the same frequencies BUT use different codes
 - i.e., Direct Sequence Spread Spectrum (DSSS), and
 - Code Division Multiple Access (CDMA)
- Two carriers: L1 = 1575.42 MHz, L2 = 1227.60 MHz
- Use BPSK modulation

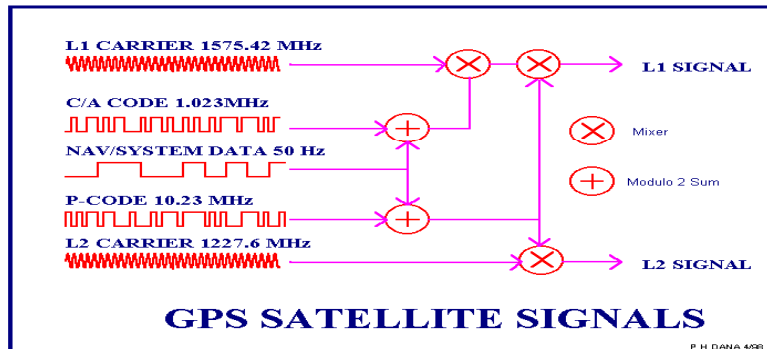


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GPS Identifying Codes



- Two types of codes
 - C/A Code - Coarse/Acquisition Code available for civilian use on L1 provides 300 m chip length
 - P Code - Precise Code on L1 and L2 used by the military provides 3 m chip length
 - encrypted P code (called Y code) provides selected availability and anti-spoofing



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GPS User



- Typical receiver: C/A code on L1
- During the “acquisition” time you are decoding the navigation message on L1
- Navigation Message - transmitted on both L1 and L2 at 50 bps
 - each frame is 1500 bits; transmitted at minute and half-minute
- The receiver then reads the timing information and computes the “pseudo-ranges”
- Obstructions to GPS satellites common
 - each node needs LOS to 4 satellites
 - GPS satellites not necessarily overhead, e.g., urban canyon, indoors, and underground
- Russia has similar system Glonass
- EU building their own system GALLIOS



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Satellite Summary



- Applications of Satellites
- Basic system structure
- Types of Satellites
- Industry in flux – move from telecommunications to data oriented applications



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